

AVIATION

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The new Pitcairn "Super-Mailwing" PA-6 climbing off the ground

VOLUME
XXV

Special Features

The Travel Air 6000

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Regarding the National Air Tour

NUMBER
9

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A scratch means - "Throw it away!"

WRIGHT Workmen do not know how to "fix" a scratch... These men, whose high technical and mechanical skill is supplemented by their feeling of personal responsibility, do not know how to remedy a defect in any finished part for a Wright engine... Their only constant when a flaw is met is to "throw it away"... A scratch however slight, a defect visible perhaps only through a microscope, is the "unlocked door" toward which stresses in the metal might strain for release.

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Wright "Cyclone"
Aircraft Engine

More Rely fly them!

WRIGHT

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ATTENDING POWER DISPLAY

Pacific Aeronautics Corp.—Los Angeles, Cal



Wright "Whisper"
Aircraft Engine

Scott Air Services, Inc.—Quebec, Rich.



The Oldest American Aeronautical Magazine

Vol. XXV

AUGUST 25, 1929

No. 9

Catapulting Airplanes

CATAPULTING airplanes off of ocean fronts has been established as a major air force to-day. The French Line for being the first to put this idea into actual practice. The Laurot or Oliver flying boat, which was well, had a launch engine loaded of some three tons, which shows that the French have progressed far in their catapulting work. Launch engines, the pilot of the plane, and its wireless and radio gear were hoisted with their plane when the Laurot was more than 400 miles from shore. As this was a forcing wind, the plane made good time and the mail carried was delivered to the post office some 15 hours ahead of the time it would have been delivered had it made the complete crossing by steamer.

The achievement is well in very worth while, but there are considerably greater possibilities to come. The course of most trans-Atlantic liners starts the Irish coast and comes within a few hundred miles of the coast of Newfoundland. The Irish coast is a day's run from the French coast, and whatever is introduced from the Irish coast of a plane is just as much extra time saved. It is estimated in two days and from New York, but it is within easy flying distance. With flying fields as complete have been established in Newfoundland and on the Irish coast a wing of nearly a day and a half could be made on the westward trip and a saving of nearly two days on the eastward trip.

The problem of flying from land to land is much more difficult. Not only does the pilot of the plane find a very small spot in a very large ocean, but wireless reporting is much more difficult at sea than on land, especially when there are local patches of fog existing. There there is the problem of what the pilot would do when he got to the ocean. Under certain conditions the steamer could be used to land the plane and create a sea with smooth enough water so that the plane could land. However, this method would not always be practicable. Of course it would always be possible to drop the mail in water right bags to be picked up by the steamer, or to have the mail bags caught by some sort of an attachment fixed to the steamer. Well, neither of these two methods would be entirely satisfactory, so it would mean that the plane that would have to return empty.

Some day these problems may be solved and when that time comes the mail will only travel by boat some seven or eight hundred miles of the eight hundred mile coast between the coast of Newfoundland and the coast of Ireland. Thanks to the work done by our navy in the development of the catapult, and to the initiative of the French Line, a new field of possibilities has been opened to the aeronautical industry.

Plumbing Troubles

AS a comparison of Atlantic flight attempts seem to be made, it is not surprising that the failure of the Captain Courtney's flight, France's flight and the Polish flight were all those surrounded by what is popularly known as "plumbing" troubles. In none of the instances was there any failure in the engine, but in all three cases the engines stopped through failure of the feed lines. The failure of France's four-engine Dornier plane was the most striking. Apparently all four engines quit at so nearly the same time that a heavy descent from a low altitude had to be made and the plane was very badly damaged. On investigation it was found that the gasometer pumping system had failed and apparently even the last of engines will not run without gasolene. It would seem as if this should have been noticed before and that an independent pumping system should have been in operation for each engine. Working out a system where each engine is independent of the other, yet at the same time can use the gasolene of any other engine is not as simple as it seems, but it certainly can be worked out.

In the case of the Polish flight the oil system failed, but as the plane was wrecked it may be hard to ascertain exactly what happened. Apparently the failure was not in the pumping system but in the gasolene system, as there was an evident and large leakage of oil. In the case of Captain Courtney the failure was due to the breaking of the main gasolene feed line. Evidently either the material used was faulty, or, what is more likely, it was so supported that it was subject to excessive strain and that failure was merely a question of time. In either case the danger of running pipelines through spaces enclosed by covering and close in the exhaust was demonstrated.

The arrangement of feed lines is not as exact science and the stresses involved are not as calculable as those on a wing or crankshaft, but none the less, it would seem as if enough experience had been gained to point out many failures from such causes. Of course, all the planes in question were in a sense experimental and the particular installation had not been tested in the time of these weeks of miles of flying which standard types run up. It is evident, however, that the arrangement of the feed lines, especially for the new types of engines which are being brought out, is a matter which requires the greatest care, and it should not be left to the shop mechanic as even to most plane designers. To the effect of the failure of a feed line is just as serious as the failure of a crankshaft, and if engine manufacturers are about the reputation of their product they should check carefully with the plane manufacturer the details of the engine accessories.

Spot Welding of Aluminum and Its Alloys

By W. M. DUNLAP

*Hydromatic, Technical Director Bureau,
American Company of America*

THE question of spot welding aluminum has been quite extensively studied in the laboratories of the Aluminum Company of America. Early in this study it was necessary to determine what was to be considered a good weld, so as to establish a standard of comparison. It was found that, and our experience has been corroborated by the work of others on other metals, that the best spot weld is one which pulls out of the welded sheet so as to leave a hole as shown in Figs. 3 and 3A. It was found that such welds could be made in 25, the Aluminum Company's designation for commercially pure aluminum, and in the alloys 35, the designation of aluminum-magnesium alloy; 17S, duralumin; 2S, aluminum-copper-magnesium alloy; and 51S, which is the designation for aluminum-magnesium-silicon alloy. It was further found that this method of determining quality is reliable as a rough test.

Fig. 1 shows the simplest form of foot-operated spot welder. The two points are shown at A and the foot lever at B. A compression spring and adjusting nut for controlling the mechanical pressure applied to the work between the electrodes during the process of welding is shown at C. An automatic electric switch, not shown in this view, is opposite to D. The regulator switch for adjusting the secondary voltage and current is shown at E. The electrode horns are water-cooled, the cooling ducts being connected by a rubber hose, shown at F.

Machines With Automatic Timing Unit

On this type of machine the time of current application is regulated by the speed with which the foot lever is depressed. Welding machines equipped with automatic timing devices are more satisfactory for spot welding aluminum.

An important difference between the spot welding of aluminum and steel is the readiness of copper to alloy with aluminum. When plain copper electrodes are used, the points are very soon coated with a layer of copper-aluminum alloy. When the electrode points become thus coated the quality of the weld is very poor and there is very great danger of burning a hole in the sheet, due to arcing. It is thus necessary to clean the copper electrode points with a file, or with emery cloth, quite frequently.

This, of course, means that the operator must spend a considerable part of his time cleaning electrodes. A recent development on which a patent has been applied for, consists in bridge chromium on the contact surfaces of the electrodes. The copper tips may be plated with chromium, or a piece of chromium rod may be inserted in the tip. With such tips, it is regular production experience to receive from 1,800 to 2,500 spots an eight gauge 25 and 35 sheet, without retouching and replacing.

The size and shape of the electrode tip is important. The diameter of the contact surface should be between .34 and .36 in., the larger dimension being for thicker sheet. The contact surface of the electrode should be slightly rounded, as shown in Fig. 2. This requirement is particularly important when welding very thin sheet. The purpose of using a rounded welding tip is to produce a spot which is uniform in shape and quality. The

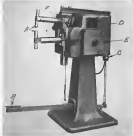


Fig. 1. Foot operated spot welder. Letter designations referred to in text matter.

rounded tip produces a dulled spot in the sheet, but if the current density and mechanical pressures are correct, the depression is not deep enough to be electrically active. It has been claimed that electrodes with hemispherical tips, as shown in Fig. 2A, will automatically reduce the current density at the electrode slots into the metal, and thus prevent excessive melting. This supposition seems of dubious merit and creates the disadvantage of producing a deeper indentation than desirable. However, it has been found that the welding of very thin gauge sheet necessitating from 0.002 to .003 in. thick, which is very difficult with flat-pointed electrodes, can be readily performed with rounded tips.

When flat-pointed, coated tips, such as are shown in Fig. 2B, are used, it is very difficult to prevent eccentric

contact between the tips and the sheet, which causes a concentration of the current at one point and results in an uneven spot. Flat-pointed contact tips will give excellent results on stock 0.05 in. or greater in thickness, but the two contact surfaces of the electrodes must be absolutely parallel at the instant of contact when making the spot. A very satisfactory method of determining whether the electrodes are in perfect alignment is to place a piece



Figs. 2, 2A, 2B. Fig. 2 is a cross-section drawing of a meter welded electrode and welding tip. Fig. 2A shows a hemispherical tip. Fig. 2B is a cross-section drawing of a meter welded electrode with a pointed contact electrode tip.

of polished metal between the tips, apply the foot pressure, and rotate the sheet through about one-fourth of a turn. If this leaves a mark showing concentric circles with a center at the middle of the tip the electrodes are properly aligned; but if the mark is off center and irregular, the relative positions of the electrodes must be altered or their tips ground by rotating a file between them with the foot pressure on until the alignment is correct. Of course, chromium plated tips must not be ground or filed.

Occasionally it is necessary to have the maximum indentation on one surface. This can be accomplished by using an electrode with a flat tip slightly larger in diameter on that side of the joint. However, the spots made in this manner will usually not be as uniformly good as when both electrode tips are of equal size and slightly rounded. Due to the greater current density required with aluminum, the electrodes should be water-cooled, and the water-cooling should be extended to within one-half inch of the contact surfaces of the tips.

As a result of the greater electrical conductivity of aluminum as compared to steel, higher current densities are necessary. Consequently, for spot welding aluminum, a machine of larger capacity must be employed than is required in the case of steel sheet of equal thickness. The principal variables requiring control are the mechanical pressure exerted by the electrodes upon the joint, the current density, and the time of current passage.

The general scheme to follow in setting up the machine for a certain job is to first reduce the mechanical pressure and current density to a point below that at which a spot will form and then to increase the current density. If "spitting" or arcing is observed, insufficient mechanical pressure is indicated. The lowest value for current density and mechanical pressure that will produce a good spot after sample welds will ascertain the most suitable setting. A satisfactory weld in aluminum is one in which the spot will tear out of one sheet, rather than one in which the two pieces of sheet mostly pull apart through the spot. Figs. 3 and 3A illustrate the failure of good spots.

No fixed rules of current, pressure and time for spot

welding different gauges of sheet of the various aluminum alloys can be given. By varying the time and pressure, it will be found that the limiting values for the current over a wide range. Judicious adjustment of each of the three variables, time, current and pressure, will allow of making a satisfactory spot weld under circumstances which apparently seem most unfavorable. In working out the conditions for spot welding a particular thickness of metal, the best rule to follow is to employ as high a current and as low a pressure as is possible without danger of arcing. The values of the current across the electrodes and the mechanical pressure recorded are only approximate. They are merely intended to give some idea of the requirements for a particular gauge of sheet.

Metal	Thickness of sheet	Amperes Secondary	Pressure in lbs.
25	0.030 in.	4000	30
	0.038 in.	3000	30
	0.050 in.	2000	30
	0.070 in.	1600	30
	0.080 in.	1400	30
35	0.040 in.	3000	35
	0.045 in.	2500	35
	0.060 in.	2000	35
	0.080 in.	1500	35
	0.100 in.	1200	35

Most metals are most easily welded when in the stage of plasticity lying between the malleable and solid states, hence these metals which remain plastic the longest while cooling are the easiest to weld, and vice versa. In order to be welded, the metal must tend to flow at some critical temperature, so that it will adhere when pressure is applied. Aluminum is one of the few metals that does not become plastic until its melting temperature is closely approached. Therefore welding conditions must be very closely controlled. Aluminum must be welded almost instantly in order to secure the desired kind of welded joint. The action for this is that reactions take place if the heating is applied over any appreciable length of time.



Fig. 3. Spot weld in 0.032 in. 25S sheet, showing four sheets welded together in one operation, and the nature of the failure when the weld is torn apart.

By making aluminum spots practically instantaneous, the weld is concentrated without time for oxidation. In fact, the quicker any kind of weld is made in any metal the better and stronger the weld is likely to be.

What has heretofore been written about spot welding is concerned mostly with the spot welding of steel. However, the spot welding of aluminum has been developed to a stage where it is used in regular production, particularly with pure aluminum and some of its alloys. Aluminum is easily welded and repairs are made in great quantities. The main requirement for a satisfactory weld is that the metal must be clean and free from dirt, or heavy

(Continued on page 395)

The Travel Air 6000

*New Type Designed for Mail, Express and Passenger Carrying I,
Powered With a Wright "Whirlwind" Engine*

PRODUCTION was started recently by the Travel Air Manufacturing Co., Wichita, Kan., on the new single seaplane, Type 6000. The plane has been designed for general commercial transport service including mail, express and passenger carrying. It is powered with a Wright "Whirlwind" J-5-C engine developing 200 hp at 1800 rpm.

The plane is a six place, externally braced, high wing type and in design and construction conforms with general sea production practice. It has a wing span of 58 ft 7 in., an overall length of 29 ft. 10½ in. and an overall height of 8 ft. 8½ in. The weight of the plane empty is 2575 lb. and the gross weight loaded is 4000 lb. It has a pay load of 944 lb. Tests have shown a high speed of 126 m.p.h., a landing speed of 50 m.p.h., a climb of 700 ft. per min. at sea level and a service ceiling of 12,000 ft. The plane has taken off fully loaded with tail wind in 720 ft. and landed under the same conditions in 300 ft. by use of brakes.

As in other Travel Air models, Type 6000 has a welded and tube fuselage and a wood wing structure. The wing is semirigid in planform and the aileron section is the Goettinger 208. Dow spars consisting of square caps with radiused plywood web are used and the ribs have square web, capspars, and truss and plywood joints. The ribs are attached by means of angle brack glued and nailed and weigh 15 lb. each. All fittings are of heat treated chrome molybdenum steel.

Skis are made in the internal drag bracing of the wing. Far bay adjacent to the root and containing fuel tanks is braced by plywood at the top and bottom and all other bays are braced by double Monowheel round

wrapped rods. Drag ribs are solid plywood webs with double spar struts between the spars at top and bottom. From the leading edge to the front spar the wing is sheathed with duralumin. The trailing edge is formed of sheet duralumin and is very light and rigid. The wing tip is formed by a good tube and two short ribs. The wing structure is built up and covered with Flightline Grade A fabric sewed through the full length of each rib.



A rear view of the Travel Air 6000.

Ailerons are of the rib type with wood structure and mounted with cast aluminum alloy hinges. Stabilizer and elevators are rectangular in shape and like fin and rudder are of steel tube construction. All control surfaces are fabric covered. The rudder is balanced but the elevators are not. The surfaces are actuated by means flexible cable and internal controls are completely inside the wing. The stabilizer is adjustable in flight. The wing is wired for landing and navigation lights and fuel lines. Travel Air blue. Lull struts are steel tubes, simply turned with dural.

(Continued on page 216)



Side view of the new Travel Air transport monoplane powered with a Wright "Whirlwind" engine

The "Super-Mailwing"

*Improved Model of Pitcairn "Mailwing" Powered with a "Whirlwind."
Has High Speed of 128 M. P. H. and Lands at 45 M. P. H.*

THE latest and largest type of plane produced by Pitcairn Aviation, Inc., the "Super-Mailwing," has just been completed at the Company's factory, Bryn Athyn, Pa., and has performed successfully in flight tests conducted by James G. Hoy, operations manager.

The Super-Mailwing is a modification of the Pressure Mailwing, which is now in use on night mail routes. It has the same wings, tail group and engine—Wright J-5-C—as the Mailwing, but has a tapered mail compartment of nearly twice the capacity of the Mailwing, 40 cu. ft. instead of 28½ cu. ft., making room for 300 lb. of mail of average bulk or 22,500 letters. This new and larger mail plane has been made necessary by the great increase in the volume of airmail carried throughout the country because of the reduction of postage rates in five cents for an ordinary letter weighing one ounce or less. Like the Mailwing, the new plane is designed primarily for mail and express, although on special jobs the mail compartment can be divided to make room for an open cockpit for two passengers, in addition to a quantity of mail.

By moving the pilot's cockpit back, reducing the space between the mail compartment and the engine and by making the fuselage slightly wider, the large increase in the capacity of the plane has been obtained with only nominal increase in the length of the fuselage. This change has, in fact, enabled an even better stream-line form for the fuselage with the result that the performance and



Front view of the new "Super-Mailwing."

maneuverability of the Super-Mailwing have proved to be as great as those of the Mailwing, PA-5. The high speed of the plane with full night flying equipment, including two parachute flares, landing lights, storage battery, etc., is 128 m.p.h. at 7350 r.p.m., with the cruising speed at 1750 r.p.m. is 110 m.p.h. Landing speed is 45 m.p.h.

The many mail and proved features of construction and design of the Mailwing are retained in the new plane. For example, the longmen and principal cross members of the welded steel fuselage are of square section tubing in accordance with long established Pitcairn practice. This material is one inch square in section with round corners and has the great advantage of automatically simplifying the welding processes and insuring better joints and greater all-around strength. No wires whatever are used in the fuselage structure, all bracing being by means of square or circular sheet tubing or tie rods welded into ribs. The fuselage is well faired over all sides with "U" section steel, a feature which contributes not only to performance but to pilot's visibility. It is also interesting to note that the covering around the pilot's cockpit is rigidly supported on a metal structure such that it will withstand the heavy usage of pilot getting into and out of the cockpit.

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A side view of the Pitcairn "Super-Mailwing," recently completed by Pitcairn Aviation, Inc.

Regarding the National Air Tour

By JOHN T. NAMI

WHAT did the 1938 National Air Tour prove as to the way of airplane reliability and efficiency?

That question has been asked this writer, and the writer knows the same question has been asked a number of the pilots who flew in the Tour, as well as several others, who, like myself, were fortunate enough to accompany it.

One of the easiest ways to answer that question is to state the facts as to what the Tour accomplished in a physical sense. That answer is simply this:

Approximately 1,300 hp., spread over 25 airplanes, transported approximately 70 men and women 6,000 mi. in approximately 70 hr. flying time.

Those are the facts. Could anything better demonstrate the reliability and efficiency of modern American aircraft? Of course, one could go further with his facts and point out that 24 of the 25 planes completed the flying time and made it to the starting point with their passengers, that the one plane forced out was compelled to do so because the pilot had difficulty obtaining parts for his Americanized foreign engine, the only one like it in the

area, the one with the foreign engine that dropped out during the early stages of the Tour? How 628 plane legs out of a possible total of 800 plane legs are scheduled time, or better, that the Tour crisscrossed 17 states, visited 32 towns and towns in 15 of them, stopped two days in sight of the cities, overcame 12 in 24 days, and returned to Detroit, the starting point, in 29 days; that two of the planes were powered by 47 hp. engines, and one of those two was piloted by a woman, that but one of the 25 airplanes were damaged in any extent during the 6,000 mi. and time, but at least (this is for the skeptics) that all of the pilots and passengers returned to the starting point in perfect health and will please with their adventures.

Of the 24 planes that completed the Tour, only three received one engine at any stage over the route. Eddie Stinson, flying entry number 20, installed one in Tulsa, Okla.; Charlie Myers' plane, Waco number 59, got one in San Antonio, and Jack Atkinson's Monocraft, number 25, received one after he had been forced down between Wisconsin and Milwaukee. Strictly speaking, none of the "new" engines installed were new. Both the "Whitcomb" placed in entries numbers 19 and 20 had been used for approximately 100 hr. each, and the Vultee engine going into Atkinson's plane had about 40 hr. against it.

Weather Conditions Generally Good

Weather conditions were, for the most part, favorable over the entire route. Except for four or five legs of headwinds during the initial half of the Tour, and two or three legs of crosswinds and squalls, very little wind of any kind was encountered. Two or three very minor thunder and rain storms crossed the path of the touring planes during the 29 days of flying, none of them having any appreciable effect on the Tour schedule.

Now, in a series of round-table reports, witness the performance of the individual planes.

Entry Number One was a Bellanca, model C11, monoplane piloted by William S. Brock, and carrying three passengers. It was powered by a Wright J5a engine. Brock's figure of merit, or score for a perfectly flown leg, was 621.4. The first 29 of the 32 legs on a scheduled time, being delayed by headwinds over the legs between Ft. Worth and Waco and Waco and San Antonio. Like most of the competing pilots, Brock was late into Springfield, Mo., due, chiefly, to a miscalculation of the distance between that city and St. Louis. The distance was estimated at 184 mi. Brock's total score was 15,445.3 points, and he landed in eighth place.

Entry Number Two, a Ford tri-motor monoplane, flown by Capt. Frank M. Hawks and carrying six passen-



John P. Wood and his "Whitcomb" entered #190 to 1941 which he won the 1938 National Air Tour.

tour, that approximately one-third of the Tour route was over mountainous, thickly wooded country, where a safe landing might not easily be accomplished; that a considerable portion of it was across unbroken desert country, where intense heat and "dead air" had to be overcome; that some of the mountainous terrain necessitated an altitude of from 8,000 to 10,000 ft., allowing a reasonable margin of safety; that temperatures from those a mile above the starting point to that of 112 deg., experienced at Tucson, Ariz., were encountered during the 6,000 mi. flight; that there were but 15 forced landings, one of these being caused by shortage of fuel and most of the remainder by minor engine difficulties; that the 25 planes (includ-

The Timm Biplane

Five Passenger, Cabin Biplane Designed to Seat Pilots in Open Cockpit
Is Powered by a 260 Hp. Menasco-Salmson

THE Timm biplane recently completed by G. W. Timm Airplane Corp., Glendale, Calif., has been given a long series of load tests at the Oakland Airport. These tests are being made preliminary to an attempt to set a new world endurance record. Otto Tamm, the designer, feels that should this airplane be successful in surpassing the endurance records which have long been held by monoplanes he will have proved that the biplane has not yet been outclassed as a long range carrier.

Flown by Captain Roscoe Turner, who will make the endurance flight attempt, the Timm biplane took off with a load of 350 gal. of fuel after a run of 900 ft. and easily attained an altitude of 6200 ft. Powered with a Menasco-Salmson 260 hp. radial engine the high speed proved to be 112 mph.

Designed for High Altitude Work

Originally designed to transport members of a Los Angeles hunting club to small high altitude landing fields in the Sierra Nevada Mountains, the Timm plane was designed with all calculations being based on 7000 ft. instead of sea level as is the custom. The cabin arrangement of the original design was unique in that it provided a semi-convertible enclosure, the rear upper half of the cabin during being convertible so that the passengers could change from a completely enclosed cabin to a large semi-open cockpit at will. The five passengers were seated in three wider chairs fastened to the floor and in two seats built into the rear of the passenger compartment, and the two pilots seated in an open cockpit (one forward of and above the cabin and provided with dual side by side control). The overall length of the fuselage is 31 ft. 9 in., the overall height in flying position is 11 ft. The span of both wings is 45 ft.

The 260 hp. nine cylinder radial Superposed Menasco-

Salmson engine is mounted on a conventional steel tube fuselage structure that is fastened to the fuselage with four bolts. The engine has been carefully covered, the cowling being neatly constructed into the fuselage which has a maximum depth of five feet and a maximum width of 44 in.

The fuselage is of chrome molybdenum steel tubing with welded joints and employs no wire bracing whatever. The forward part of the fuselage, from propeller spinner to just forward of the pilot's cockpit, is covered with the



Side view of the new Timm Biplane

aluminum cowling. The entire cabin is constructed of plywood, and the remainder of the fuselage is covered with fabric. The plywood covering is secured to the fuselage frame by specially rounded square struts that grip the tubing and to which the plywood is glued and nailed. It is said that this plywood construction eliminates the unpleasant creaking noises that are experienced when fabric is used alone. A large baggage compartment is located ahead of the passenger cabin. (Continued on page 97)



Front quarter view of the Timm Biplane powered with a 260 hp. Menasco-Salmson

Completing Fine Le Roy Airport

New York Millionaire Pile
Bought Friendship Operating
Fleet of Planes

LE ROY, N. Y.—Duluth is to be one of the first fields in the State, a 30-acre airport is being completed here off the main Buffalo highway in the vicinity of the village of Donald Woodard, millionaire air enthusiast. Woodard, who has a fleet consisting of four C-54's, Waco's, a Beechcraft power four-engine plane, and a Cessna Skylark, owned by a C-4 engine, is now in Europe, where he is making arrangements to bring the Friendship Flighter airplane, Friendship, which he recently purchased, back with him to Le Roy.

When the new hangar, shops, office, and fuel are completed in shape, a full recreational program will be inaugurated at the Woodard airport by the D. W. Friend Service, Inc., of which Woodard is president. C. H. Powell, Hamilton, former pilot, is vice-president of this company, while Leonard Hendrick is secretary and H. Kirk Truitt, Treasurer.

Completing Large Hangar

"An outstanding feature," declared Captain Hamilton is an airplane, "will be a private school, which will be complete in every detail. Complete instruction in aerodynamics, navigation, and mechanics will be included in a four-year program," he said. "While the regular flight instruction at the Waco will be followed by cross-country work in Cessna's, two of which he has already received. The Cessna Skylark, on the other hand, is to be used for engine training."

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School to be Feature

The hangar, being built under Hamilton's direction, is 20 ft. by 100 ft. It is fitted with concrete and has a 60 ft. concrete apron outside the building doors. More than a dozen airplanes of ordinary size can be housed here.

Next to the hangar, a lot has been reserved morning the entire length of the building and surrounding in a storage tower. The ground level of the lot for storage will be 200 ft. on an elevation, the second floor will be built into a real room with stairs and built into

Nebraska Cities On Continental

LINCOLN, NEB.—Lincoln, as well as Ashland, Newark, York, and Aurora, will be on the regular transcontinental aircraft route by October 1, it was recently reported by Captain F. C. Hedges, director of airports for the Department of Commerce, who was in Lincoln for a short time.

Rebuilding of the route, and preparation of the landing field, business, and other necessary equipment is to begin immediately.

Arrangements for women pilots and visitors, and the top floor will be a pilot's classroom with large windows overlooking views of all parts of the field. A hangar is to be installed at the apex of this tower.

An Engine Warning Device

In the town there will be a complete machine shop, a stock room with a couple of radiators, and a small office. The field, on airplane repair room, and a locker and shower room for pilots. There will be a special hot air system in the engine room for warming engines during the winter months preparatory to morning take off. Snow here, by this system will be piped under the turbine engine of an engine. Another feature will be a temporary grandstand for the motor.

One feature, a large mail flat, with air being made in the town, is flying the Woodard Friendship plane at present in a passenger airplane service. It will act as a courier to the new school.



Donald Woodard, who heads the D. W. Friend Service, and Vice President Rexford Hamilton.

To Dedicate Sunbury Airport September 1-3

SUNBURY, Pa.—Committee to mark the dedication of Sunbury 120 acre airport will take place September 1-3. An invitation is extended to all pilots and planes who can be present for the dedication.

The program will open on the evening of August 31 with a banquet and entertainment, on the Wall Street Garden for all guests, sponsored by the Sunbury Flying Club, Chapter of Commerce and Commerce Club. On September 1 the field will be dedicated to Wesley L. Smith, superintendent of Eastern division, National Air Transport. These ceremonies will be accompanied by Army aviation, army, navy, flying, and lunch for visiting pilots. The evening will be devoted to a display of fire works and the entertainment at guests. On September 2 and 3 the program will include aviation, plane, radio, and building exhibits and dinner for guests at Sunbury State Valley County Club.

Sunbury Airport has accommodations for all men and crews of planes. A new hangar has been erected and electrical and water pipes have been installed. The airport is 100 ft. long and 100 ft. wide, and 100 ft. north and south.

Long Island Field Dedicated

ROCKVILLE CENTER, L. I.—Sunbury Airport, a field comprising 100 acres in the northeast corner of town, has been dedicated. The field, which is known as the "Old Field," has been given to Rockville Center by Robert W. Long, headmaster of the town. More than 400 guests attended the dedication.

More than 400 guests attended the dedication.

Minneapolis-Duluth Line Uses Seaplanes

MINNEAPOLIS, Minn.—One of the most complete passenger lines in the United States is now operating out of this city following the recent establishment of seaplane service between Minneapolis and Duluth, Minn. by Northern Air Lines, Inc., a division of the International Air Lines System, Inc. The new line is being operated by a variable landing field in Duluth to land in the case of an emergency at any time. The line is being operated by the Minneapolis-Duluth line. The line is being operated by the Minneapolis-Duluth line. The line is being operated by the Minneapolis-Duluth line.

The planes had left out from Lake Superior, it is being operated by the Minneapolis-Duluth line. The line is being operated by the Minneapolis-Duluth line. The line is being operated by the Minneapolis-Duluth line. The line is being operated by the Minneapolis-Duluth line. The line is being operated by the Minneapolis-Duluth line.

Committee on Marking Soon to Make Reports

WASHINGTON, D. C.—The work of the National Airports Marking Committee, which was set up by the National Airports Marking Committee last May, is now being completed. The committee is now making a report on the work of the committee. The committee is now making a report on the work of the committee. The committee is now making a report on the work of the committee.

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Expend Park Airport

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Building Seaplane Landing

LOS ANGELES, CALIF.—Construction has begun on a seaplane airport at the mouth of the San Gabriel River, near Los Angeles. The airport is being built on a site of 100 acres. The airport is being built on a site of 100 acres. The airport is being built on a site of 100 acres.

Oakland Reports Now Teletyped

OAKLAND, CALIF.—Four telegraph machines have been installed in the airport between Oakland and the radio station at Concord, Calif. By remote control, these machines type out the weather reports and other flying information available to mail and passenger plane pilots.

Midwest Firm Opens New Field at Aurora

MONMOUTH, ILL.—The Midwest Airplane Corp. has opened a new flying field at Aurora, Ill., and will operate the field for the company. The field is being operated by the Midwest Airplane Corp. The field is being operated by the Midwest Airplane Corp. The field is being operated by the Midwest Airplane Corp.

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To Dedicate Burke Airport

OKMULGEY, CALIF.—A new hangar field at Okmulgey is being offered to the local chamber of commerce as a new field for the city. The field is being offered to the local chamber of commerce as a new field for the city. The field is being offered to the local chamber of commerce as a new field for the city.

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Springsfield-Montreal Service

SPRINGFIELD, MASS.—Wentworth & Ryan Airways has been awarded a contract to operate a service between Springfield, Mass., and Montreal, Canada. The service is being operated by the Wentworth & Ryan Airways. The service is being operated by the Wentworth & Ryan Airways.

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Reviews

Book Review: "The Airplane" by John D. Smith. The book is a review of the airplane industry. The book is a review of the airplane industry. The book is a review of the airplane industry.

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A Personal History of the American Airplane Industry

The book is a review of the airplane industry. The book is a review of the airplane industry. The book is a review of the airplane industry. The book is a review of the airplane industry. The book is a review of the airplane industry.

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Aviators' Pliers

The influence of aerodynamics on the design of small hand tools is indicated in a set of three special pliers now offered by the Borgert Steel Products Company, Newport, Penn., under the name of the "Aviator Kit."



Special Aviators' Kit

Formed in the group is a general use plier of austenitic design with a long thin nose brought to a finished point. Its chief features are hand-fitted curves that cut with ease, deeply rolled gripping teeth and end grip, adjustable joint for expanding the gripping jaws and "recessed grip" non-slip handles. The length of this plier is 7 in.

For working in confined spaces necessitating to minimize pliers, the machine will find the 6-1/2 in. long special nose pattern a helpful adjunct. This pattern is also fitted with curves and is deeply rolled in the extremity of the jaws for securing a tight hold. The handles and jaws are spring tempered.

The diagonal cutting pattern, shown in the illustration, measures 6 in. in length. They are light in weight, but are built for heavy duty work. This special Aviator's Kit, weighs only 16 oz.

Dot Grease Pump

TO MEET the demand for a machine capable, hand operated grease pump the Dot Lubrication Division of the Carr-Patterson Co., 31 Ames St., Cambridge, Mass.,



Pump opened and ready to be filled.

has developed and placed on the market the Dot Little Giant grease pump. The new pump has a capacity of five lbs., weighs only 23 lbs., filled, and develops pressure

up to 10,000 lbs. It is equipped with Alenox, Zerk and ball bearings.

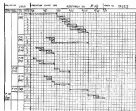
One of the interesting features of the Dot pump is its ability to develop unusually high pressure with little effort on the part of the operator. The entire body of grease in the tank is held under constant pressure by a heavy spring working between the pump cup and a plate attached to the bottom of the pump tank. By a simple rock and pivot mechanism the grease is drawn through a screened opening in the plate and delivered through a series of cylinders to the Scanner cylinder at the top of the pump. These by the operation of a booster piston, a pressure of 18,000 lbs. may be developed

Specifications of the Labor

(Continued from page 597)

board clerk who takes his ticket for the job just completed and stamps it with the time stopped. He then advances to the top pocket one copy of the order for the next job to be done, and gives the other to the workman, first stamping down with the time started. He also gives the workman the drawings, instruction sheets, etc. that go with the job. On the original copy of the ticket for the work finished the quantity produced will then be entered by inspector or welder, and it will be sent to the office to be noted, stamped, checked with payroll, and posted to costs. The duplicate copy from the control board may be kept for a time by the clerk as a record of jobs finished each day.

The amount of work ahead of any machine is shown by the tickets in its pocket on the board. The total estimated time required to perform this work is obtained by adding up the estimated times of all those tickets. While due to errors in judgment, the estimates may be high or



Production chart for industrial operations.

low on individual tickets, these errors will largely balance each other if a considerable number of tickets be taken. For the determination of work ahead, therefore, the accuracy of exact time studies is not essential. The judgment of the superintendent or foreman, or of a man familiar with the work, will give very good results.

When such a plan has been put in operation it is commonly found that some machines or work places will have less time work ahead of them, while others will have a

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great accumulation. These latter are the "bottle necks" which hold back the flow of production. Sometimes study of the neglected operations shows that they are not being done efficiently. On these occasions time study brings its greatest and quickest returns, and here is where it should be started.

If new work is scheduled against a machine which already has much ahead of it, either the new must wait its turn or else work already scheduled must be delayed and set back to wait room for it. If this latter be impossible then the new job will just have to wait. If this complete new job has many parts and operations it will frequently happen that while many of them can be put through promptly a few will take longer or will be held back by the completion of one of two machines and will delay the whole assembly. Sometimes work partly completed on the assembly floor has to be stopped because some minor part is not at hand and cannot be produced promptly. The situation of such delays requires not only the scheduling and dispatching of work by the control board, but the preliminary planning of it as well. This means that the ordering date of each part shall be set as an indication to the date it is required that, whether from stock, purchase or manufacture, every part shall come to hand when it is needed, neither too soon so that it lies around and gets lost or injured nor too late so that it delays the assembly.

The detailed planning of production is like a game of chess, every piece on the board. Its possible moves must be kept in view. It is also like the planning of town measurements over a network of branches, offings, and junction points on a railroad system; the crossings, sidings, and meeting points of all trains must be in the dispatcher's view. An expert chess player can visualize an board and make his plan's final result. A train dispatcher requires some wide experience of the conditions on his division.



Control board hand used by the Greater-Aircraft Assembly Co., Inc., Hawthorne, Md.

and sees it through his diagnosis of concerning parts and train movements. The planner of production requires similar help. The nature of this help depends on the character of the production. If it is simple standard and continuous, little is required. If it is discontinuous and complicated, it may be desirable to have a very complete graphical planning system covering all parts and assemblies, with constant check-ups and corrections.

Here, as before said, the airplane and engine industries will occupy an intermediate position. As the parts be-

come more standardized, and are produced and stocked in larger quantities, the final assemblies will become more mere matters of taking and assembling finished parts from an adequate stock. Also, the production of those parts can be arranged more and more in large lots and regular sequences on equipment adequately provided for the job.



Production chart for continuous operations.

me only is rare. The assembly will then require but little special planning. It will go through a steady stream as automobiles do now. And then the quantity of any part becomes great enough so that a specific group of machines is kept in continuous operation to produce it, very little planning will be required for it. However, under such different levels and lots of parts required on specific dates, are made as limited orders which need coordination each other in production and follow each other over the same machines a slight interference control plan may well be required.

On standard operation assemblies the planning may consist of filling in with the specific data a standard chart showing at what stage of the assembly each elementary part is required and the length of time that part is allowed to produce it. These data cannot be filled in at free will and only after comparison with the conditions of work ahead, as shown by the control board, and after such changes and rearrangements of the work scheduled for other orders as it may be necessary to make. Here is shown the closer most cases in. On special case production work no such standard plans are available; special agencies have to be made and the work is even more difficult though not unmanageable. In the meantime, still, however, it will be increasingly possible to make use of standard production charts.

In the heavy industrial lines, this chart shows graphically the time required to perform each operation on each of the free parts, two sub-assemblies and the final assembly of an article No. F-10. These production times and the starting dates of each part are so arranged as to bring all parts to completion just when conditions in ideal condition not always easy to attain. For standardized work and those charts without order numbers or data are prepared in advance by letterplotting or letterprinting. When an order is to be scheduled one of the ideal charts are taken and after reference to work ahead as shown by the control board, the date when each operation should be started is filled in on it. The pre-written labor tickets are then

scheduled on the control board to be run as the data are. By the use of dotted or colored lines parallel and adjacent to the full heavy lines, and representing the actual daily production as the order, in terms of several days work, such a production chart may also be run, showing the actual progress made on each operation. A complete picture of the shift of orders on the order is thus presented.

Charts like this are most useful for planning and controlling production, where complicated processes are put through in isolated lots. But for parts regularly made and coming through in a steady flow, so that the detailed preliminary planning is not called for, only the daily and cumulative progress need be presented. The charts will take a form similar to that just shown, but only the current actual production on each operation will be given.

To illustrate the difference, this chart has been drawn to cover the same assembly and parts as in the preceding example, but for continuous instead of occasional production. This is only one of several ways in which these facts might be presented. Simple continuous tabulations of production, in parallel columns, might be used, and would involve less work than charting, but are less satisfactory. Also, more complicated combinations of lines and symbols in different colors have been devised. Since the whole production system if but one part fails behind, the essential thing is to make evident whether all parts and operations are marching along as step of the pre-arranged plan. For this, the lines illustrated are well adapted. However, it may be pointed out that these methods will be ineffective unless used and followed up by a wide-awake production department.

This completes the consideration of methods of production control. In the next article the forms and uses of labor taken will be taken up.

The Titan Biplane

(Continued from page 355)

In the cabin is by means of a large door on the left side of the fuselage. Two large windows equipped with sliding glass are provided on each side of the cabin. A large door opening into the baggage compartment is provided on the outside of the fuselage on the left side.

The pilot's cockpit is centrally mounted, all controls and instruments being most conveniently arranged. Control surfaces are operated by wires passing over pulleys and jacks. All wires in the tail pass down and under the passenger aisle between the floor and the lower. Although not carrying landing lights or flares as standard equipment, the plane is equipped with navigation lights and a lighted instrument board. The pilot's vision is excellent due to the location which is quite high and is well back from the engine. Sun and wind do not interfere with the pilot's vision. In landing the wheels are visible from the pilot's seat.

While the normal gasoline capacity is provided by two 35 gal. gasoline tanks situated in the center section of the upper wing, for the purposes of the endurance flight the cabin has been fitted with additional tanks giving a total fuel capacity of 712 gal. A 40-gal. oil tank is mounted forward of the pilot's cockpit and directly below the engine.

The wing arrangement is that of a single bay biplane. Both wings have a span total span of 53 ft. and the same chord which is 6 ft. They are rigged with a gap of 6 ft. 9 in. and a stagger of 216 in. There is no noticeable dihedral, the angle of incidence being one degree for the lower and two degrees on the upper wing. The lower wing is built in two panels each of 20 ft. 2 in.

spars, which are joined in the fuselage at the lower long ribs. The upper wing is built in three panels, two of 18 ft. 6 in., and a center section of eight feet, which is of plywood covered.

The single X strut is located at 15 ft. 6 in. from the center on each side and the wire bracing consists of lift and landing wires coming in a vertical plane, all drag wires being taken within the wings themselves and to the fuselage by the X struts at the center section. Double MacPherson strut wire are used for all external bracing.

The wing structure is conventional, employing box spars with square caps and two-ply 45 deg. Duralumin ribs, while the ribs are of spruce in Warren cross structure.



Rear quarter view of the Titan Biplane

ture with the six struts reinforced by plywood fenders to resist buckling and to keep the true shape of the curves. Drag bracing within the wing is single wire with compression members are of steel tube built in so as to take the tension between the spars.

There are four ailerons of wood construction. They are mounted in pairs by a single strut between each pair and are operated through the lower wing. A trailing edge feature of the aileron control is the mounting of an auxiliary aileron above, and rigidly connected to each lower aileron by two short struts. Serving to balance the ailerons for center control, these will lift warping as they do so comparatively unaided, as the main ailerons are the lateral control at low speeds for they add an appreciable amount of surface to the total of the four ailerons, each of which measures 9 ft. by 20 in.

The landing gear is a T-tail development of the rubber-hydraulic type. Four short struts extend from the lower fuselage structure to the ends of the straight axle of heavy bias tension steel tubing. The two rear struts contain shock absorbing units built by Ciro Tross, which employ both rubber and controlled liquid shock absorbers to ease the shock of landing and to check the rebound. The two forward struts are rigidly attached to the axle but are hinged to the fuselage. They are rigidly connected with struts. They provide a well braced, sturdy landing gear which is free to move back and up against the action of the absorber units as the airplane meets the jolts incident to ground maneuvers. The tail end is of conventional shock tube, a piston and is mounted with woven rubber chord absorber unit.

The entire empennage is constructed of chrome molybdenum steel tubing. The rudder is unusually large and well balanced. The fin is braced in the horizontal stabilizer by two cross struts. The stabilizer is braced to the fuselage by two fixed struts in front and is adjustable through the two rear struts by means of a worm gear drive operated from the cockpit and which is built to function only once under maximum stress. The elevators are also braced for ease of control and are operated by means of wiring within the fuselage in a lever control that extends outside the body near the tail and connects to the Rudder

horn by means of a rigid steel tube, thus eliminating outside wires in the vicinity of the tail surfaces.

Although complete flight tests have not yet been given this airplane it has shown a maximum speed with full load of 312 m.p.h. and will load at 35 m.p.h. Having taken off with 2500 lb. load after a run of approximately 500 ft. there is evidence that the designer's hope of hitting a 7500 lb. load will be realized.

Although this model was a special job constructed for high altitude work, it will probably be placed in production if Captain Turner is successful in setting a new endurance record. At the present time the G. W. Tress Aircraft Corp. is developing all factory facilities to the quantity production of a small touring monoplane.

Specifications of the Titan Biplane as supplied to Aviatron by the manufacturer:

Length overall.....38 ft. 9 in.
Height overall.....11 ft. 11 in.
Aurion.....11 ft. 11 in.
Span (both wings).....45 ft. 6 in.
Chord (both wings).....6 ft. 9 in.
Gap between wings.....15 in.
Area wings, total.....530 sq. ft.
Area of ailerons.....54 sq. ft.
Area of horizontal stabilizer.....26 sq. ft.
Area of elevator.....30 sq. ft.
Area of fin.....33 sq. ft.
Area of rudder.....15 sq. ft.
Power plant.....360 by 9 cylinder Napco-Salmon High speed.....117 m.p.h.
Cruising speed.....100 m.p.h.
Landing speed.....35 m.p.h.

Regarding the National Air Tour

(Continued from page 354)

port, was equipped with three Wright "Whitewards" showing figure of eight with 280 ft. and he landed just over 36 m.p.h. of the 32. The most engine on Hawley's plane suffered a broken crankshaft on the flight from Indianapolis to St. Louis and although the pilot had to fly 100 m.p.h. into St. Louis with two engines, he arrived on schedule.

Regarded with the St. Louis-Springfield leg Hawley was late for six consecutive legs, due chiefly to the headwinds over that route. After leaving San Antonio he flew the remaining legs of the route on scheduled time. Hawley's score was 24,500 ft. He finished second.

Entry Number Three was a Tress Air biplane powered by a 135 hp. Fairchild-Camper engine. J. Nelson Kelly was the pilot and he carried a mechanic. Kelly also flew 26 of the 32 legs in perfect form. He received a new engine, a first that had been on the Fairchild-Camper powered Waco Number 28 at Indianapolis and completed the Tour with it. Carburator trouble caused Kelly to be late starting from the field in Tucson, and as a result he was late into Miami. He also served late into Portland, principally because of having to fly approximately 20 mi. out of the way in following the valley route via Bessberg on the flight from Portland. The remainder of the Tour was made with perfect service, although he was forced down at Portland, Wash., to replace two blown sparkplugs. Kelly's plane, and that flown by Gould Reed, introduced the Fairchild-Camper engine to many sections in the West. His total score was 13,700 ft. and he finished third. Kelly's figure of merit was 322.

Entry Number Four was an Eaglehawk, flown by

"Benny" Howard and powered by the Napco-Salmon. After scoring two perfect schedules of 316.6 points each over the first three legs, Howard was forced down between Wichita and Tulsa. This was due to engine trouble and inability to secure engine parts quickly enough he dropped out of the contest.

LAST "Roller" Daily GX-3 Entry

Entry Number Five, a Carter "Roller," piloted by "Doc" Robertson, failed to show up on all of its eight legs, principally because at an engine that was understood to have considerable hours to its credit before the Tour started. The plane is powered by an OX 5, and it was the only OX 5 job in the Tour. Robertson flew 10 of the 32 legs on scheduled time. The figure of merit was 237 ft. and he carried one passenger. Number Five was late into San Francisco because of time trouble on the field at Fresno. While attempting to start up the engine preparing to leave San Francisco, the propeller struck Robertson's mechanic on the arm and was broken. Because of the delay in replacing the broken propeller Robertson's leg was down the next three stops. One more perfect leg was flown, then again trouble plagued him again. He returned to the field in Fresno, where his plane was left by Spokane. At Santa, N. D., on the leg between Miami and Fargo, Robertson's engine again forced him down. After arriving in St. Paul, Number Five flew the remaining five legs on schedule. His Robertson's score was 5,949 ft. and he finished fourth.

Entry Number Six, Dave P. Levy, pilot of the Shamrock entry, was the only pilot flying the Tour who scored his figure of merit over every leg of the route. Levy's perfect score for a leg was 366 ft. and there was not a weak leg in record. He was turned down at once, at Thompson Falls, Mont., on the leg between Spokane and Missoula, but managed to get in on time. Levy also was the youngest pilot on the Tour, being but 31 years old. His Shamrock was powered by a "Whiteward" and he carried one passenger. The score on Number Six was 12,700 ft. He finished seventh because of his low figure of merit.

Entry Number Seven, this plane, a Bell "Anasir," was piloted by Arthur Graham. It was equipped with a "W" motor and Graham flew 26 of the 32 legs on scheduled time. Handwheels and the navigational distance between St. Louis and Springfield, caused Graham to be late on practically all of his longer leg legs, although he was late arriving at the airport before the start from Chicago, and was late into Battle Creek for this reason. Graham's figure of merit was 5429 ft. and he carried two passengers. He finished eighth on the Tour. His score was 13,327 ft. Finished ninth.

"Anasir" Flies 29 Perfect Legs

Entry Number Eight, Louis G. Mosier piloted Entry Number Eight, a Bell "Anasir" airplane, powered by a "Whiteward" engine. He flew 29 of the 32 legs on time, maintaining in doing so a required speed of 109.38 m.p.h. Mosier's figure of merit, 5024 ft. was broken between St. Louis and Springfield, for reasons stated above, between Ft. Worth and Waco because of severe headwinds, and between Waco and Dallas, when he was forced down at Round Rock, Tex., after a valve spindle had broken off. Mosier had one other forced landing, between Tucson and Spokane, when he landed to replace a blown piston. This landing, however, was not a forced one. Mosier's figure of merit was 5024 ft. and he finished ninth. His score was 12,938 ft. and he finished ninth.

Entry Number Nine: "Whitcomb" powered Ryan brought, piloted by E. W. "Bud" Cleveland. "Bud" flew 41 of the 32 legs on schedule, being affected with a wide variety of malaise. His figure of merit, 2903.3, was raised on the first leg. Cleveland made it on the second, missed again on the third and fourth and made it on the fifth. On the sixth leg between Tulsa and Ft. Worth, he was forced down when his aircraft stalled. A landing on a small and rough field resulted in a considerably damaged undercarriage and broken prop. Headquarters issued several other important legs until the Tour arrived in San Diego. Cleveland was late starting from San Diego. Cleveland was late starting from San Diego because of having to make a telephone call. In Los Angeles he broke his tail skid and was late into Fresno. Three perfectly flown legs followed, then started trouble caused Number Nine to be late into Portland. Five more perfect legs then, in fact, Miami, the Ryan, was late and caused Cleveland to be late into Miami. The seventh leg legs were flown on scheduled time. Cleveland carried three passengers. His total score was 11,034.6 and he finished twentieth.

Entry Number Ten: Al Blaney piloted Entry Number Ten, another "Whitcomb." Ryan brought, making 27 of the legs on scheduled time. He was one of the three Tour pilots who managed to get into Springfield, Mo., on schedule. The following day, however, between Wichita and Tulsa, Blaney had misfortune of experiencing engine trouble caused by engine trouble. His score for this leg, 2493.4, was the lowest score he made, his perfect score for a leg being 3446.5. Headquarters gave him schedule for 14. More than 1000 miles of engine trouble on those legs between Tulsa and San Francisco caused the rest. After leaving Tulsa Blaney flew his Ryan over every leg on scheduled time. He carried two passengers. His score was 17,728.9 and he finished seventh.

Fully to Start

Entry Number Eleven: A Challenger, model C-4, biplane, powered by a Fairchild-Caterpillar engine. Never started.

Entry Number Twelve: Number Twelve was an Eagle-Rock, Haco "C" powered, and flown by Floyd P. Cleveland. Cleveland carried two passengers and flew 36 of the 32 legs on schedule. Headquarters issued practically all of Cleveland's important scores. He flew every one perfectly after leaving Vero. Cleveland's total score was 11,442.7. He finished eighth.

Entry Number Thirteen: Number Thirteen was a Mohawk, "Pinto," powered by a Warner "Scout" engine, could not be made ready in time to start.

Entry Number Fourteen: This entry, a Travel Air biplane, was the only plane in the tour powered by a Ryan-Caterpillar (C) engine. After making a very nice start, flying 15 of the first 16 legs on schedule, the engine died. The biplane was then part of the tour, George B. Beck, the pilot, became the victim of accessory trouble in El Paso and was 11 days catching up with the remainder of the Tour. Maguire's trouble, experienced on the full at Port Hope, El Paso, prevented Beck from taking off with the other boys and flying into his home town, Tucson, Ariz., with them. It was a tough battle for the little pilot and all of the stress was laid upon him. Beck again was late starting from the Tour again in Great Falls, Mont. Although after postponing the Tour Beck flew the remaining eight legs on schedule, his trouble in El Paso proved costly, and he finished twenty-second. His figure of merit was 304.6 and his total score, 6,712.4. Beck carried one passenger over each of the 5,600 mi.

Entry Number Fifteen: This entry, a Pinnerwing composite, with two "Whitcomb," was to have been flown by C. C. Harford, of Detroit, but it could not be made ready in time to start.

Entry Number Sixteen: Vance Brown, who introduced the Ryan M-14 to many parts of the United States in 1935 when he flew Ryan Number 23 in the National Air Tour of that year, again piloted a Ryan, this time a biplane. Vance carried two passengers, and flew 28 of the 32 legs with perfect scores. Headquarters and the trans-Mississippi between St. Louis and Springfield caused Vance's late perfect legs. After leaving San Antonio he flew the rest of the route on scheduled time. His figure of merit was 302.4 and his total score, 18,360.3. He finished in ninth place, a position he had held from the start.

Sooner's Self in Tulsa

Entry Number Seventeen: A Seafarer, powered by a Jugh, this plane started with Jay Seawright at the controls, but was lost to Clarence W. Dehn of Elgin, Ill., while the Tour was in Tulsa, after which Merle V. Janssen became pilot and Helen his one passenger. The plane was flown over the 32 legs with only one being imperfectly flown, that being the aforementioned St. Louis-Springfield leg. The Seafarer's figure of merit was a total score of 11,608.5 and a figure of merit of 428.5.

Entry Number Eighteen: John Paul Wood, of Umanu, Wis., flew Waco No. Number Eighteen, in first place in the Tour having accumulated a total score of 29,111.4. Wood's figure of merit, 803.5, was the highest figure of merit among all of the competing planes. Wood took the lead on the first leg and held it throughout the long flight. Except for the St. Louis-Springfield leg he flew the entire Tour with perfect scores. The prime importance of a low start and undercarriage trouble on the Waco was noted 4 seconds late and his undercarriage 3.1 seconds. The pre-Tour tests on Waco's Waco changed as follows: Gross Weight 4600 lb; Empty Weight, 1570 lb; Content, 140, 580 lb. 74 per cent of the Capacity, Load 712 lb; Maximum Speed, 126.1, 161 per cent of the Maximum Speed, 150.8; and Engine Displacement 280. Wood won the Tour by a wide margin. He carried Frank Charters as a searunner, and experienced no trouble with his plane and Wright J4.

Broken Pairs Pairs Missed

Entry Number Nineteen: Number Nineteen was another Waco Type, this one piloted by Charlie Miller. Miller carried four of his own legs on the 32 legs, being late on the St. Louis-Springfield leg and being late in Dodd Field, near San Antonio, with a broken wheel. Fortunately, San Antonio was a two-day stop, and through the sponsorship of Jack Pratt, San Antonio took over the leg. Miller was able to get a new engine. Although Miller's score for the Waco San Antonio leg was reduced from 718.2, his figure of merit is 12.8 because of his forced landing, he took off on scheduled time from San Antonio, and experienced no more late arrivals. With a total of 22,129.1 he finished fourth. He carried two passengers. His plane had a J4.

Entry Number Twenty: Eddie Strawn, who was the Tour's last year and who has flown the last three Tours, took fifth place this year, with his "Whitcomb" powered by a Warner "Scout" engine. Like most of the first fifteen was late into Springfield, but he managed to get perfect scores for the rest of the legs, except that between Tulsa and Ft. Worth. Eddie replaced his engine in Tulsa, thus had the "new" engine introduced at San Antonio. He

carried four passengers, and his total score was 21,243. Strawn's figure of merit was 675.1.

Entry Number Twenty-One: was another Strawn, that one being a Warner equipped Stinson Junior and flown by Randolph G. Page. Page had the only Warner engine in the Tour. His first 26 of the 32 legs on schedule, his principal delays being caused by headwinds over the early portion of the route. After leaving San Diego Page flew the remaining fourteen legs without any difficulty arrival. With a total score of 22,796.5, he led the class of the other ten Strawn competitors by more than 1,500 points and finished third. His figure merit was 733.8 and he carried two passengers.

Entry Number Twenty-Two: Headquarters on Ft. Worth-Waco leg and that much amounted St. Louis-Springfield leg caused the only two imperfect scores piloted by Bruce Hiram pilot of Stinson-Douglas Number Twenty-Two. Bruce, whose plane was powered by a "Whitcomb," had a figure of merit of 630.3. He narrowly missed being late into St. Paul because of a blown wing, which caused him to return to the Fargo airport, shortly after taking off on that leg. With a total of 19,671.5, he trailed Eddie Strawn, and placed sixth. Bruce carried two passengers.

Entry Number Twenty-Three: Like the Seafarer, Entry Number Twenty-Three, the Lockheed "V" was flown over the Tour with two different pilots at the controls. Bob Cantwell piloted the plane through its pre-Tour tests and then flew to Los Angeles, but there was something to withhold him from the flight. He was late in the St. Louis-Springfield leg. He was late in the rest of the route. After three legs missed at the start, Cantwell succeeded in finishing his perfect score of 973.7 over the Springfield-Wichita leg. He was delayed by trouble into between Ft. Worth and Waco, but once started perfectly well he was relieved in Los Angeles. Scheuchner flew the Lockheed with perfect scores over all of the remaining legs, except two, being delayed by tree trouble on the Fresno field and again in Santa Cruz by its failure to return on the Chicago leg to meet the pre-Tour test. He finished third. The Lockheed was powered by a J4C and carried two passengers. Altogether it finished the route with 25 perfect scores. Being one of the two oldest planes in the Tour, Lockheed's performance was a long-term triumph ahead of the remaining crews. Scheuchner's total score was 17,935.6. He finished seventh.

Pears Pilot Fought Monoplane

Entry Number Twenty-Four: Richard D. Pears, pilot of the Fairchild monoplane, Number Twenty-Four, flew 29 of the 32 legs with perfect marks, two of his delayed arrivals being due to headwinds and the other to the mis-scheduled changeover from the St. Louis and Springfield legs. After leaving San Antonio Pears flew over every leg on schedule. The Fairchild was equipped with a Wright "Whitcomb." Pears carried one searunner, and over his trip, Tulsa to Ft. Worth, carried an extra searunner in the person of Arthur C. Hunsantrant. It might be mentioned here that a number of the competing pilots were generous enough to carry supplies weight over portions of the long route whenever they were asked. Shortage of space in the touring plane made this regular service, at least in the early part of the tour. Pears' total score of 15,854.4. He finished twelfth in the Tour.

Entry Number Twenty-Five: Number Twenty-Five was a Bellanca flown by Capt. George W. Haldeman. Haldeman carried no passengers, but he was from Indianapolis and was well known, was held back by headwinds between Ft. Worth and Waco, and was delayed getting into Fargo by delays to his plane while attempting to take off at

Mont. but he flew 26 of the legs on scheduled time. While waiting for a take-off at Great Haldeman's plane struck a tree just as it was about to leave the ground. This prop was bent and the under portion of the fuselage broken. Haldeman, however, was delayed only long enough to bring his plane from 462.2, his figure of merit, down to 377.4. The delay was not caused by a broken prop and, carried, broke the rest, these passengers had a total score was 15,850.4 and it finished fourteenth.

Hawkeye's Main Gun Shooting

Entry Number Twenty-Six: In many ways, Entry Number Twenty-Six, the Hawkeye, flown by Mrs. (Phyllis) Pratt, gave Oke, of Memphis, Tenn., gave the most satisfying demonstration of any plane in the Tour. This monoplane's enclosed cabin monoplane, powered by a 47 hp. V8 engine, was the only one to fly over the entire route. The highest maneuvers, starting from the airport, were carried out way through the strongest headwinds, and despite an accident on the Fresno field in Arizona, Tex., that kept it separated from the Tour plane for six days it managed to fly 15 of the 32 legs on scheduled time. Except for the portion between Tulsa and San Francisco, Number Twenty-Six was piloted by Mrs. Oke. Between El Paso and San Francisco the piloted Entry Number Twenty-Six, carried Mrs. Oke. The plane was practically perfect. It led to that of Number Twenty-Six, but lost to that of Monoplane Twenty-Nine, gallantly placed last place at the disposal of Mrs. Oke, and returned to Tulsa from El Paso to repair the damaged Number Twenty-Six, and by a lack into the Tour. The woman pilot had flown from Tulsa to El Paso in the pre-Tour flight piloted by Hacks. Mrs. Oke ground looped the one plane while landing on the Army field in Tulsa. The flight around the Tour made by the Hawkeye, was the longest and most difficult flight made by any plane there was in this country. With a total of 3,525.2, Mrs. Oke finished last, but that did not prevent her flight from being the most surprising, enlightening and extraordinary feature of the 1938 Tour.

Entry Number Twenty-Seven: Number Twenty-Seven was entered in the Tour as a Cariboo-Two monoplane, powered by three Anzani, but the plane never appeared at Ford Airport for the pre-Tour tests.

Entry Number Twenty-Eight: This plane was a Waco Type, powered with a Fairchild-Caterpillar and piloted by M. Gould faced. Despite the headwinds and a goodly portion of engine trouble, Gould flew the Tour with 20 perfect scores, his figure of merit being 321.6. In Indianapolis, as already related, the prop was taken off Number Twenty-Eight and placed on Number Thirteen. This last Tour separated from the Tour and it reached Tulsa, Oklahoma, on the 29th of August. The plane's portion in Memphis, Tex., a constant, took place through a heavy engine crank over, in Spaulding field returned to the field with a broken engine stand, and in Fargo it was two hours late starting because of a broken motor arm stand. Because of its various troubles, Gould, no doubt, became the lastest working pilot on the Tour, and he deserves a lot of credit for finishing twenty-eight with a total score of 7,890.6. He carried no passengers.

Entry Number Twenty-Nine: As previously said, Jack Hunsantrant, Entry Number Twenty-Nine, a Monoplane, powered by a V8 engine. Number Twenty-Nine was flown from El Paso to San Francisco by Mrs. Oke, and accumulated 14 perfect scores over the 32 legs of the Tour. Hunsantrant was from Indianapolis and was well known, was held back by headwinds between St. Louis and Waco, was delayed getting into Fargo by delays to his plane while attempting to take off at

down again at Laguna Beach, Calif., but this time it was being flown by Mrs. Orelie. Adkinson was into Comrag, Calif., due to visiting on the San Francisco airport for Mrs. Orelie, who did not choose to take off in the Vig. This caused her to be late on the three legs between Fresno and Portland. In Portland an illness caused Adkinson to be late in covering the next four jumps. He resumed the Tour at Great Falls, however, and requested his figure of merit, 352.1, for the five successive legs. The Vigle engine then dropped a valve on the Wisman-Milwaukee leg, and brought Adkinson down. He replaced the engine with a new one, and again resumed the touring places in Chicago. Mohair Adkinson ran Mrs. Orelie carried Mrs. Eddie Stron from Ft. Worth to Marfa, and again from San Diego to Los Angeles, and Adkinson carried American's representative over two legs of the Tour between St. Louis and Wichita.

Although we could never get any of the pilots flying the Tour to admit it was the most remarkable fact of the 1928 Tour was the splendid way in which all of the pilots navigated their planes over the huge and trying terrain. Several of the younger men, and, we understood, Mrs. Orelie, prior to the Tour, never had a whole lot of experience flying cross-country. Despite this handicap, very few cases of being "lost" were reported. One or two of the pilots were equipped with such modern navigating instruments as earth inductor and gyroscopic compasses, although most of them depended solely upon ordinary compasses and hand-McNally maps. Over a large portion of the route, namely, over the desert and mountain country, where such convenience guides as waterways were lacking, maps were virtually useless. Several of the pilots referred their compasses affected by what was believed to be a natural deposit in the mountains of West Texas.

The highly varied temperatures encountered on the Tour had little or no effect on the engine, all a majority of the pilots are to be believed. Thus, they said, met with on some of the fields caused them more trouble than did the extreme heat.

The "Super-Mailing"

(Continued from page 593)

Streamlining has been applied wherever possible—mainstream lights are now so made the wings and radials with celluloid covers of proper color; wing wires fitted to fittings inside instead of outside the wings, the springs being covered with celluloid windows for inspection; the streamlining covers with their added resistance and the necessity of removing them for inspection. The celluloid meshed map is made into the streamlining engine covering giving an extremely clean appearance to the engine installation and further reducing head resistance. A small gap between the meshed and the fuselage covering provides adequate and even cooling by vacuum action.

The under carriage of the new plane is very similar to that of the PA-5, having, however, a slightly wider track. The undercarriage design is such that the pneumatic shock absorber struts are rectangular along the best shock makes and this applies to the wheels and bridle also.

As in the Mailing PA-5 the most important of the Super-Mailing is in the fuselage between the engine and the pilot's cockpit and centered around the center of gravity of the plane so that whether a heavy or light load is

carried in this compartment the center of gravity is not moved more than one or two inches. The balance of the plane and its flying characteristics are, therefore, little affected by taking on or leaving off large loads at various stops along an aerial route.

The Super-Mailing is fully equipped for night and blind flying with compass, turn and bank indicator, speed indicator, altimeter, tachometer, oil pressure, oil temperature and fuel gauges, magnetron lights, landing lights, battery, and indirectly heated instrument board with rheostats to dim or brighten the instrument board lights in day or moonlight nights. All wiring is sheathed in metal cables and all metal parts of the plane, including



View of one of the wing brace wire fittings and the streamlined upperwing cowling used on the Plesner "Super-Mailing."

wing fittings, are electrically bonded to eliminate the possibility of an electrical discharge due to lightning, and no ground shielding in case radio is used. Space is provided for the installation of radio beam equipment.

Orders for fifteen Super Mailings, totaling over \$150,000 were received by Plesner Aviation, Inc., before the first plane was completed. Among these ordering places are the Departments of Commerce, National Air Transport, Colonial Air Transport, Colonial Western Airways and Canadian Colonial Airways. The plane will also be used on the New York-Albany and Atlantic-Miami aerial services of Frontier Aviation, Inc.

The general details and specifications of the Plesner Super-Mailing PA-5 are as follows:

Overall wing span, upper wing 30 ft.
Overall wing span, lower wing 30 ft.
Overall length 22 ft. 10 in.
Overall height (standing on ground) 9 ft. 2 in.
Wing chord, upper wing 9 ft. 2 in.
Wing chord, lower wing 4 ft. 8 in.
Wing aspect ratio 7.42
Wing area 252 sq. ft.
Dihedral, upper wing 1 in 4
Dihedral, lower wing 1 in 4

Cabin Interiors

Progressive manufacturers are turning to mohair fabrics for the finishing of aeroplane cabins. The use of this fabric for upholstery and trimming gives a luxurious appearance to the interior without adding materially to the cost or weight of construction.

Visit our booth (Number 9) at the Aeronautical Exposition in Los Angeles, Cal., during the week of September 8-16. Representatives will gladly show you samples of mohair fabrics particularly adapted for the interior of aeroplane cabins.

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AIR MEET
AT DISSEMINATION OF
WILMINGTON, DELAWARE AIRPORT
BELLANCA FIELD
OCTOBER 6th, 1928

STANDARD TIME 10:00 AM AT 10:00 AM. THE AIR MEET WILL BE HELD AT 10:00 AM. THE AIR MEET WILL BE HELD AT 10:00 AM. THE AIR MEET WILL BE HELD AT 10:00 AM.

THE AIR MEET WILL BE HELD AT 10:00 AM. THE AIR MEET WILL BE HELD AT 10:00 AM. THE AIR MEET WILL BE HELD AT 10:00 AM. THE AIR MEET WILL BE HELD AT 10:00 AM.

THANK YOU for watching AVIATION

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Gap at base line 62 in.
 Sagger 22 in. 42 lbs.
 Area of airframe 21 sq. ft.
 Area of stabilizer 15.5 sq. ft.
 Area of elevator 34.5 sq. ft.
 Area of fin 4.2 sq. ft.
 Area of rudder 0.4 sq. ft.



A view of the engine cooling of the Pieten "Super-Melting" showing the exhaust manifold.

AVIATION

August 23, 1938

Weight, empty 1840 lb.
 Total weight 3000 lb.
 Wing load 11.5 lb./sq. ft.
 Power loading (220 hp) 13.6 lb./hp.
 Cruise speed 70 m.p.h.

The first plane, which performed entirely satisfactorily without alterations is being very completely and thoroughly (as the New York-Arlene-Melrose army in both day and night flying without carrying mail.

The Travel Air 6000

(Continued from page 592)

wood and covered with fabric, the rear one on each side being adjustable.

The welded steel tube fuselage is braced in the forward end by tubes and in the rear by winged struts. Canvas are mounted with plywood and the entire structure is covered with Gracie A. Flightex fabric and finished in Travel Air blue. Two doors in the right side of the fuselage give access to the cabin. The door at the rear opens into the rear passenger compartment and the door at the forward end opens into the pilot's compartment. Provision has been made for four passengers and two pilots seated side by side. Comfortable wider chairs and ample space for storage of baggage are provided for passengers. A triplex windshield affords excellent visibility for the pilot and all other windows are of plate glass and may be rolled up or down. The cabin is finished in fine velour and all wood and metal parts are lacquered. A dome light in the passenger's cabin and heaters in both passenger and pilot's compartments are furnished. All instruments required by the Department

AVIATION

August 23, 1938

of Commerce are installed and adequately illuminated. The right flying equipment is standard and includes a full set of navigation and heading instruments. Dual controls are provided for the pilot who or who by side and wheel brakes are operated by auxiliary tanks. All vital points in the control system are accessible by ropes in the lateral centering.

The power plant is readily accessible by removing the cowling. The engine mounting is detachable and is made of welded steel tubing. A manifold ring developed by the Travel Air Co. comprises the exhaust system. Two



Rear quarter view of the Travel Air 6000, designed for general transport service.

alternating fuel tanks, each having a capacity of 35 gallons, are built into the wing. A fuelstar gasoline strainer is provided and can be drawn down to the pilot's compartment. All fuel lines are 1/2 in. copper tubing. The engine is fitted with an Delphi hand inertia starter and a standard steel propeller.

Landing gear is of the divided axle type with a 9 ft. wheel and 30 x 5 floats which are broken. Travel Air also shock absorbers are used as compression members.

The tail unit is of the conventional non-removable type employing rubber discs as compression.

The specifications furnished by the manufacturer are as follows:

Length overall 30 ft. 10 1/2 in.
 Height overall 8 ft. 8 1/2 in.
 Aircraft section Geringer
 Wing span 36 ft. 2 in.
 Chord 78 in.
 Wing area 382 sq. ft.
 Area of airframe 20.4 sq. ft.
 Area of stabilizer 15.5 sq. ft.
 Area of elevator 34.5 sq. ft.
 Area of fin 4.2 sq. ft.
 Area of rudder 0.4 sq. ft.
 Weight empty 1840 lb.
 Power loading 13.6 lb./hp.
 Cruise speed 70 m.p.h.
 High speed with full load on sea level 80 m.p.h.
 Landing speed 50 m.p.h.
 Take off run (fully loaded on wind) 720 ft.
 Landing run (No wind, with brakes) 300 ft.
 Climb at sea level 700 ft. per min.
 Climb to 3000 ft. 12 min.
 Climb to 8500 ft. 32 min.
 Service ceiling 12,000 ft.
 Absolute ceiling 15,000 ft.
 Gasoline consumption at cruising speed 12 gal. per hr.
 Gasoline capacity 70 gal.
 Range at cruising speed 630 mi.
 Endurance at cruising speed 6 1/2 hr.



This Lad is in your keeping

YOU, who employ this fine, expanding American lad to fly your planes, have shouldered a responsibility.

Yes, it is his skill that will win through, but it is your foresight in the matter of providing reliable equipment that enables him to devote his entire attention to flying. The Boyce MotoMeter Aviation Type is a reliable motor test indicator. Regardless of altitude, or of atmospheric conditions, it furnishes the pilot dependable information on motor performance.

The motor is the heart of an airplane. Unless the heart functions smoothly the lad who is in your keeping, is "out of luck." Give him reliable instruments!

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Boyce MotoMeter, Aviation Type, made exclusively of non-ferrous materials, accurate and dependable.

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SEAPLANE



by **Edo**
 AIRCRAFT
 CORPORATION
 College Point, N. Y.

Standardized All Metal
 Seaplane Floats

spots was 1,380 lb. The average diameter of the spots was $\frac{3}{16}$ in. The failure in every case was due to the spots shearing. The average breaking load for the S15W specimens, joined by means of two spots, was 275 lb. When the S15W strips were joined by means of three spots, the average breaking load was 1,250 lb. In this case the strips failed, the fracture occurring through the

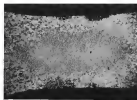


Fig. 6 Longitudinal section of spot weld through two points of S15W sheet from which the oxide film was removed previous to welding.

width. The average diameter of the spot welds in the S15W was about $\frac{1}{16}$ in. The average strength of the spot welds in the S15W strips was about 10,000 lb. per sq. in. of weld. The calculated value for the 17S in the cast condition is 10,000 lb. per sq. in. The average strength of the spot welds in the S15W strips was 14,500

lb. per sq. in. This compares with an estimated value of 11,000 lb. per sq. in. for 17S in the cast condition.

The results of these tests would seem to limit, for the present at least, the use of spot-welded joints in structures to any extent for certain purposes, such as in the construction of aircraft, particularly as stressed parts. Spot welding of duralumin, however, may have possibilities in place of riveting on covering and other parts not subjected to severe stresses. The spots could be placed close together and heat treatment could be applied, subsequent to welding, for the purpose of increasing corrosion and improving the strength of the joint.

Figs. 8, 9, 30 and 31 illustrate a few design cases where spot welding may be substituted for riveting. Fig. 9 shows an aluminum part with the handle spot welded on. Four of these parts with the handles spot welded, and four with the handles riveted, were broken in a compression test. The loads were applied so as to duplicate, as nearly as possible, the conditions that would exist if the parts were fitted with some substance, and were picked up by the handles. The handles failed under the following loads:

Comparative Strengths of Riveted and Spot Welded Handles

Style of Fastening	Load required to break	Remarks
Spot weld	104 lb.	Chk. failed.
Two rivets	215 lb.	Handle sheared.
Two rivets	104 lb.	One rivet sheared, one torn out.
Two rivets	169 lb.	Both rivets torn out.
Average	407 lb.	
Spot Welded—two spots	420 lb.	Both spots torn out.
Spot Welded—two spots	203 lb.	Both spots torn out.
Spot Welded—two spots	445 lb.	Both spots torn out.
Spot Welded—two rivets	363 lb.	Both rivets torn out.
Average	427 lb.	

The average difference in strength of the two kinds of joints is not large enough to condemn spot welding, especially when it is considered that the strength of either joint is greater than would be needed for the use to which the part would be put. However, it shows that spot welding does not yield to riveting by any broad margin when the question of mechanical strength is considered.

Fig. 11 shows an aluminum article that is assembled by spot welding. Over 100,000 of these bands have been spot welded during the past few months. These bands were formerly joined, the rate being 250 bands per day, at a cost of \$54.72 per 1,000 pieces. At the present time 750 bands per day are being welded at a cost of \$39.58 per 1,000 pieces. This shows a very substantial increase in production as well as decrease in cost. There are many articles which could be manufactured from aluminum on a practical scale by means of spot welding. These include filing cabinets, radio shells, metal furniture, various sorts of containers that need not be water tight, cooking utensils, etc., etc.

A Common Practice

Spot welding of steel is very common practice among manufacturers of sheet metal objects. A riveting is usually effected in cases where it is possible to apply spot welding in lieu of other forms of welding or joining. Where it takes the place of riveting, the economy appears in eliminating the cost of the sheet themselves, and in time saved in the process. Spot welding is much faster than riveting, since there is a total absence of drilling, alignment of holes and the spacing of the rivets.

Spot welding accomplishes in one operation the same work for which riveting requires three and sometimes four, namely, piercing, counter-sinking and inserting the rivets.

The spot welded joint is virtually as strong as a riveted joint. It remains so even if the rivets (which) to clean, we can it become loose after a period of service. This last is an important item. Aluminum rivets have a tendency to

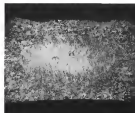


Fig. 7 Longitudinal section of spot weld made on two points of S15W sheet without removing the oxide film.

to stretch and allow some "play" in developing between the parts joined. The spot weld, even if it does stretch, is integral with the two parts, and cannot possibly develop any play. There are many cases where spot welding may be substituted for riveting or autogenous welding,

Commander Byrd Chooses Paramount Tanks



The two large gas tanks shown at the left are those chosen by Commander Richard E. Byrd for his South Pole flight. They are built of aluminum with all joints, seams and rivets welded.

Paramount Aluminum Tanks Increase Your Pay-Load

Aluminum lightweight tanks carry about 265 pounds of gas per pound. PARAMOUNT tanks have passed Army, Navy, and Civil Aeronautics specifications. They are used by all prominent aerialists throughout the country.

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THANK YOU for aviation AVIATION

SIDE SLIPS

By ROBERT R. OSBORN

Mr. R. L. R., of Paterson, N. J., sends in a clipping from a New York paper of an article describing a special ship which will be entered as the race to the Pacific Coast. Part of the description says: "Black-headed mackerel in the Rockaway Field, it is said that the motor registered a speed of 250 miles an hour." Mr. R. L. R. inquires, in the note accompanying the clipping, if this isn't a speed record for engine black mackerel. We haven't been able to consult the N. A. for official speed testing records but can state that this speed has been recorded unofficially. We were witnessing a motor black test a few years ago when something or other in the motor failed to operate as expected and we saw that the test vessel was making more than 250 m.p.h. when it went through the steel rod.

There is another interesting clipping from Mr. H. N. T., of Philadelphia, whose initials should be easily recognized by all who know anything about aviation and air races in Philadelphia. Mr. H. N. T.'s letter says, "Please find enclosed clipping from the August 21st issue of the Evening Bulletin. This would appear to be a notice for the members of your Society for the Education of Newspaper Editors. It, as the picture indicates, the member is starting a loop at about 500 ft. from the ground, the use of the anti-collapse gun in such a case would seem unnecessary except in an attempt to make 'insurance' doubly sure." The picture shows a bombing plane flying upside down over a group of National Guardsmen operating an anti-aircraft gun. It is evidently one of those composite pictures, which have been named "composographs" by the New York tabloids. So, we herewith add a "composograph" picture to the aeronautical chronicles section. As soon as we can find a member to join the society, we'll investigate some of these complaints.

A recent issue of the "Domestic Air News" of the Department of Commerce has some interesting statements to the effect pertaining to parachute jumping, from which we quote: "Parachute jumping, except in emergencies for the purpose of saving life, is placed in the category of acrobatic flying." As to our reaction to parachute jumping is considered, but emergency jumps are for the purpose of saving life.

A further interpretation of the rules by the Department states "The jump must be made and parachute opened above fifteen hundred feet." No penalty for landing that audience is mentioned and we suppose this is because the penalty is self-applying if the jump and opening is made at a much lower altitude.

We see by the newspaper that another airplane has been blown from the deck of a boat for the first time in history. There has been a fairly good crop of them first fights in history from a ship that year, that being about the fifth since Chamberlain "proved it could be done."

A photograph of Colonel Lindbergh was broadcast by radio from an airplane in a recent test of a new invention. This was an excellent photograph to select for this first time as there would be millions of people who have been wondering what the young chap looks like.

Curtiss Leadership

Throughout the world, the Curtiss system of group engineering and modern method of production is known for its accomplishments. One has but to mention the leading service types of military aircraft in our own Services to measure the degree of its success.

Airplanes

THE CURTISS HAWK

A single-seat fighter and advanced training plane, and its sister ship, the Sea Hawk, for Naval carrier services.

THE CURTISS FALCON

Two-place observation and attack ship, used by the Air Corps, U. S. Marines and the National Guard.

THE CONDOR

Latest type of Air Corps bomber with its great performance, now under construction for the U. S. Air Corps.

THE FLEDGLING

The result of design competition throughout the industry, selected for primary and advanced training by the U. S. Navy.

Engines

THE D-12

Standard high-performance engine of the services.

THE CURTISS "CONQUEROR"

600 H.P. direct drive and geared.

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